



1	Airborne polarimetric Doppler weather radar: Trade-offs between various
2	engineering specifications
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8	Abstract NCAR/EOL is investigating potential configurations for the next generation airborne phased
9	array radar (APAR) that is capable of retrieving dynamic and microphysical characteristics of clouds and
10	precipitation. The APAR will operate at C-band. The APAR will use the electronic scanning (e-scan)
11	feature to acquire the optimal number of independent samples for recording research quality measurements.
12	Since the airborne radar has only a limited time for collecting measurements over a specified region
13	(moving aircraft platform ~100 m/s), beam multiplexing will significantly enhance its ability to collect
14	high-resolution, research quality measurements. Beam multiplexing reduces errors in radar measurements
15	while providing rapid updates of scan volumes. Beamwidth depends on the size of the antenna aperture.
16	Beamwidth and Directivity of elliptical, circular and rectangular antenna apertures are compared and radar
17	sensitivity is evaluated for various polarimetric configurations and transmit/receive elements. In the case of
18	polarimetric measurements, alternate transmit with alternate receive (single channel receiver) and
19	simultaneous reception (dual channel receiver) is compared. From an overall architecture perspective,
20	element level digitization of transmit/receive (T/R) module versus digital sub-array is considered with
21	regard to flexibility in adaptive beamforming, polarimetric performance, calibration, and data quality.
22	Methodologies for calibration of the radar and removing bias in polarimetric measurements are outlined.
23	The above-mentioned engineering options are evaluated for realizing an optimal APAR system suitable for

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24 measuring the high temporal and spatial resolutions of Doppler and polarimetric measurements of 25 precipitation and clouds.

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Keywords— radar; phased array; airborne; sensing; Doppler, polarization; microphysics; weather; digital; analog; design; architecture

29 1 Introduction

30 Characterizing location, intensity, and motion of hurricanes, tornados, and extreme precipitation events,

31 and understanding effects of clouds and aerosols on the earth radiation budget requires a better

32 understanding of the kinematic (storm motion and structure) and microphysical processes (particle growth,

33 phase changes) within these storms. This remains a challenge for both the scientific and operational

34 communities. Observing these events is challenging using ground-based radars as the conditions that lead

35 to high-impact events typically occur in remote areas (e.g., hurricanes over the ocean, orographic

36 precipitation in rugged terrain) or because of large uncertainties on the timing and location (e.g., tornado,

37 extreme precipitation events) related to sub-optimal radar coverage. Airborne radar is a powerful tool to

38 observe weather systems, in particular, storms over complex terrain, the ocean, polar regions, and forest

39 regions not easily observable by ground-based radars (Bluestein and Wakimoto, 2003). A scanning Doppler

40 radar on an airborne platform is used for estimating dual-Doppler winds with the help of rapid scanning as

41 the aircraft flies past a storm (Hildebrand et al. 1996). Scanning Doppler radar with dual-polarization

42 capability on an airborne platform is capable of measuring dual-Doppler winds and retrieving particle types

43 (ice or water) and shapes, and liquid/ice water contents using reflectivity (Z), differential reflectivity (Z_{DR}),

44 specific propagation phase (K_{DP}), and linear depolarization ratio (LDR).

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46 At present, no other instrument other than an airborne polarimetric Doppler phased array radar system has
47 the potential to estimate high temporal and spatial measurements of 3-D winds and microphysics





48 concurrently (Vivekanandan et al. 2014). NCAR's Earth Observing Laboratory (EOL) is currently

49 conducting a design study for a future airborne phased array radar.

50 Between 1992 and 2013 NCAR operated research quality Doppler radar, ELDORA/ASTRAIA (Electra 51 Doppler Radar/Analysee Steroscopic par Impulsions Aeroporte, hereafter referred as ELDORA). The 52 ELDORA was configured with dual slotted waveguide array antennas using dual-transmitter, dual-beam, 53 rapid scan and step-chirped waveform (Girardin-Gondeau et al. 1991) that significantly improved the along-54 track spatial resolution from 750 m to 300 m when compared to NOAA's airborne tail Doppler radar (TDR) 55 (Hildebrand et al. 1996). The ELDORA was jointly developed by NCAR and the Center de Recherché en 56 Physique de l'Environment Terrestre et Planetaire, France. It collected research quality Doppler and 57 reflectivity measurements that continue to set the standard for airborne radar; however, ELDORA X-band 58 radar's penetration into precipitation is limited by attenuation and it is not designed to collect polarimetric 59 measurements to remotely estimate microphysics. ELDORA has been placed in dormancy because its 60 airborne platform (Naval Research Lab P-3 587) was retired in January 2013. The US research community 61 has strongly voiced the need to continue measurement capability similar to that provided by ELDORA 62 (Smith et al. 2012).

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71 The combination of remote and *in situ* sensors on a single airborne platform will serve the observational 72 needs for broader scientific communities of cloud microphysics, mesoscale meteorology, atmospheric 73 chemistry and climate, and it will fill a critical gap in the current airborne observing facilities. APAR 74 deployed on a long on-station time aircraft, such as the C-130, will allow investigation of weather systems 75 such as monsoons, tropical cyclones, severe convection over continents, orographic precipitation, 76 convection over the oceans, and polar and low to middle atmospheric chemistry. A schematic of the APAR 77 antenna panels on the C-130 is shown in Figure 1. APAR will feature four removable C-band active 78 electronically scanned arrays (AESA) mounted on top, both sides, and the bottom of the aircraft. Each 79 antenna will have dual-Doppler and polarimetric capabilities. This configuration, when integrated with data

Figure 1. Notional schematic of APAR AESA antenna panel placement on the C130. There are two sidepanels on port and starboard of the fuselage aft of the rear personnel doors.





from the NSF/NCAR C-130 nose radar will provide 360-degree horizontal surveillance radar scan coverage and volumetric data collection. Radar products can be displayed in real time. The AESA placement provides for flexible scanning capabilities for 3-D data volume generation.

83 The real-time radar displays on the aircraft provide situational awareness to aircraft pilots allowing for 84 safe aircraft operations in the vicinity of extreme weather. The 3-D volume-scan data not only can help 85 guide the NSF/NCAR C-130 research in and around weather of interest, but also has the potential to be used 86 to guide other aircraft conducting research in the vicinity. The C130 is a versatile and capable research 87 platform that carries a wide variety of scientific payloads. The C130 has a 10-hour flight endurance, a 2,900 88 nautical mile range at up to 27,000 ft., and a payload capacity of up to 13,000 lbs. NCAR EOL/RAF 89 maintains the NSF/NCAR C130 aircraft in its fleet for airborne atmospheric measurements, including 90 dropsonde, in situ sampling and remote sensing of clouds, chemistry, and aerosols.

91 This paper is organized as follows. Section 2 describes radar system, and its major sub-systems. Rationale 92 for selection of transmit frequency is presented in Section 3. Discussions related to polarimetric 93 configurations and advantage of agile beam scanning is presented in Sections 4 and 5. Various antenna 94 aperture configurations and corresponding beamwidth characteristics are presented in Section 6. The 95 sensitivity of the radar measurements depends on transmit and receive hardware characteristics, polarimetric 96 measurement configuration, and signal processing. Expected radar sensitivity as a function of a few key 97 parameters is discussed in Section 7. PAR with digital architecture is amenable for consistently maintaining 98 data quality, deployment of radar with repeatable and robust calibration and also formation of fan-beam and 99 pencil beam configuration for imaging rapidly changing weather system. In this regard possible analog and 100 digital beamforming architecture configurations are compared in Section 8. A brief description of 101 calibration requirement is discussed in Section 9. Estimates of biases in radar measurements as a result of 102 cross coupling in polarimetric mode is shown in Section 10. Section 11 presents a summary.





104 **2. System Description**

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106 Preliminary design specifications of peak power, beamwidth, dual-polarimetric configuration, scan timing 107 sequences and signal processing are outlined in Vivekanandan et al. 2014; some key technical specifications 108 are presented in Table 1. The APAR will operate at C-band. It will use the e-scan feature to acquire the 109 optimal number of independent samples for achieving 1 m/s accuracy in radial velocity, 1 dB in reflectivity 110 and 0.2 dB in differential reflectivity accuracies with a sensitivity of -11 dBZ at 10 km. Since the airborne 111 radar has only a limited time for collecting measurements over a specified region (moving aircraft platform 112 ~ 100 m/s), beam multiplexing will significantly enhance its ability to collect high-resolution, research 113 quality measurements. Beam multiplexing reduces errors in radar measurements while providing rapid 114 updates of scan volumes (Weber et al. 2007, Yu et al. 2007).

115 From an overall architecture perspective, element level digitization of T/R module versus digital sub-116 array has to be carefully considered with regard to flexibility in adaptive beamforming, polarimetric 117 performance, calibration, and imaging of rapidly moving weather system. For achieving desired sensitivity 118 and range resolution, pulse compression is proposed, with a compression gain of at least 15 dB. However, 119 this will require transmission of short pulses for covering the blind zone created by the strong pulse at the 120 expense of overall radar sensitivity due to long and short pulses. A staggered pulse repetition frequency 121 (PRF) technique for extending Doppler Nyquist interval was extremely valuable for the ELDORA is 122 considered, but its evaluation is beyond the scope of this document.

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124**Table 1.** Technical Specifications of C-band APAR

Parameter	Numeric value
Operating Frequency	C-band: 5.35 - 5.45 GHz (FAA requirement)





Antenna Aperture	38" major and 35" in minor radius ellipse.
Maximum full panel thickness (radome + antenna + T/R modules and MMIC + heat sink and mounting/support frame	<= 9 inches
Maximum weight for each AESA assembly	<= 450 pounds
-3dB Beamwidth	< 2.2° (broadside on Tx)
Sensitivity	-11 dBZ at 10 km with 0 dB SNR
Reflectivity Variance	<1 dB
Doppler Velocity Variance	< 1 m/s
Produce full polarimetric matrix	Z, V, W, Z _{DR} , LDR, Φ_{DP} , $ ho_{HV}$
Calibrated Z _{DR} for particle shape and QPE	Z _{DR} <= 0.2 dB
Differentiate liquid and ice	LDR < -22 dB
Differentiate melting	LDR < -27 dB
Collect uncorrupted weather data near surface	Within 400 m of surface, <-10 dBZ at 5 km range
Polarization Tx and Rx	H or V linear

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127 It is common for phased array radar antennas to be modular in design. Figure 2 shows a conceptual block 128 diagram of the radar system. The radar block diagram is divided into two main parts: the front end and the 129 back end. The front end consists of all parts of the phased array which reside outside the fuselage of the 130 aircraft and is represented by items: (i) the radio frequency (RF) array antenna front end, (ii) T/R modules, 131 (iii) array antenna backplane in Figure 2. The RF array antenna front end includes low-profile dual-132 polarization antenna element radiators, a power distribution board, and a data distribution board. The radar 133 back end consists of all parts of the system that reside inside the aircraft cabin and is represented by items: 134 (i) radar digital back end, (ii) radar processor/display, and (iii) radar scheduler in Figure 2.







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Figure 2. Simplified block diagram of PAR. It consists of six components: (1) array antenna front-end, (2)
T/R modules, (3) array antenna backplane and (4) radar back-end (5) radar processor/display and (6) radar
scheduler

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The desired goal is to meet or exceed the current sensitivity and spatial resolution of the ELDORA radar while adding polarimetric capability. The ELDORA is a four-frequency system and hence produces nearly four times the number of independent samples as a conventional, single frequency radar. To match the reflectivity and velocity variances achieved by ELDORA in the same time interval, the pulse compression is





144 used. This requires that APAR bandwidth (Tx and Rx) needs to be increased by a factor of four and four 145 range gates averaged. Also, to match the sensitivity of a high peak power radar, it is generally recognized 146 that to accomplish this with solid-state power amplifiers, the use of pulse compression is essential. When 147 using pulse compression in order to recover the close-in range obscured by the longer, compressed pulse, 148 short, single frequency pulse(s) must also be used. For APAR two such "filler" pulses at different 149 frequencies must be transmitted each pulse repetition time (PRT). This is necessary to have sufficient 150 independent samples to reduce reflectivity and velocity variances while still achieving the desired along-151 track resolution.

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153 The T/R module consists of a matrix of individual T/R modules. It is a multi-layer printed circuit board 154 that includes monolithic microwave integrated circuit (MMIC) components, namely, digital phase shifters, 155 attenuators, a power amplifier (PA), low noise amplifier (LNA), various drive amplifiers (DA), high-speed 156 field effect transistor (FET) switches, and a field programmable gate array (FPGA). The digital 6-bit phase 157 shifter supplies the necessary phase to the RF signal for steering the beam in a specified direction. The 158 digital attenuator tapers transmit and receive amplitudes across the active aperture for reducing antenna 159 sidelobes and also for aligning the amplitudes of each T/R element. Transmit and receive amplitudes are 160 tapered across the aperture to lower sidelobes. Various aperture-tapering options are presented in Section 7. 161

The high power amplifier (HPA) amplifies the RF transmit signal and is capable of handling short and long pulses. It dissipates a significant amount of power and the heat generated must be removed for safe operation of the T/R module. The HPA has the shortest lifetime among all of the components in the T/R module due to its self-heating. Heat sinks are placed outward into the space between the antenna aperture and backplane printed circuit boards. Cooling can be accomplished via forced air convection or a liquid cooling via a cold plate. The LNA determines the noise figure of the receive chain and it is placed closest to





the receive port of the antenna. Each T/R element is coupled to a two-port radiator for dual-polarization transmission and reception. An FPGA controls the attenuator, phase shifter, and switches for the desired performance of APAR.

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The radar backend includes master FPGAs for communicating scan angle, polarization, and pulse information to the FPGAs in each line replaceable unit (LRU) from the radar scheduler. The radar back end consists of up and down converters between intermediate frequency (IF) and RF, digital transceivers, and a host computer for generating radar measurements including mean velocity, spectrum width, and dualpolarimetric observables and associated displays for all generated parameters.

3. Transmit frequency

178 X-band and shorter wavelengths are preferred choices for mobile ground-based and airborne deployments 179 because the radar antenna is much smaller in size than the S-band or C-band antennas for a specified beam 180 width. The larger backscatter cross section of particle sizes smaller than the wavelength (Rayleigh 181 scattering) significantly improves the detection limit (Lhermitte, 1987; Clothiaux et al. 1995). In the 182 Rayleigh scattering regime, the radar cross section at X band is 10 and 20 dB larger than at C band and S-183 band, respectively. For a specified transmit power and antenna gain, X-band radar is also more sensitive in 184 detecting lighter precipitation than S- or C-band radar. However, a major disadvantage of the X-band radar 185 signal is that it is more susceptible to attenuation than an S-band and C-band radar signal. The intervening 186 precipitation between radar and measurement volume causes attenuation. The amount of attenuation is 187 proportional to the intensity of precipitation. Horizontal X-band co-polarization reflectivity (Z_{HH}) and 188 differential reflectivity (Z_{DR}) are usually underestimated due to attenuation of the radar signal as it 189 propagates through precipitation. Attenuation in precipitation at X-band is about a factor of five to seven 190 times larger when compared to C-band (Bringi and Chandrasekar, 2001).





In order to characterize the effect of attenuation on S, C and X band polarimetric radar measurements, intrinsic reflectivity and attenuated reflectivity were simulated for a rain event. The rain medium is characterized by a drop size distribution (DSD). Polarimetric radar measurements at S-band are used for estimating DSD (Brandes et al. 2004). The data used in this study was collected in east-central Florida during the summer of 1998 in a special experiment. For the retrieved DSDs, reflectivity and attenuated Z_H in S, C and X band polarimetric were simulated using rigorous electromagnetic scattering cross-sections (Vivekanandan et al. 2004).

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200 Intrinsic reflectivities without considering the attenuation effect were almost similar in S, C and X as 201 shown in Figure 3a, b and c except there are subtle differences due to Mie scattering at C and X-band. 202 However, effects of attenuation on reflectivity are noticeable in Figures 3e and 3f. In the regions south of 203 the radar, X-band reflectivity is severely attenuated and the precipitation is not detected. In comparison with 204 X-band, the C-band reflectivity is weakly attenuated. In this example, the convection system produced 205 about 0.04, 0.4 and 2 dB/km of attenuation at S, C and X band, respectively. The maximum total 206 accumulated attenuation could reach about 1.2, 10 and 30 dB for S, C and X band, respectively. X-band 207 radar is subjected to the most significant attenuation in precipitation systems and the S-band radar has the 208 least attenuation. The C-band radar signals experience less attenuation when compared to X-band radar 209 signals.







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Fig. 3: (a), (b) and (c) indicate the reflectivity without attenuation simulated from DSD for S, C and X band polarimetric radar in Sep. 17, 1998 2211UTC, respectively. (d), (e) and (f) indicate the measured reflectivity with attenuation effect for S, C and X band polarimetric radar, respectively. The radar is located in the center of each panel.

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However, attenuated C-band radar measurements must be corrected for attenuation effects before retrieving rain rate and microphysical information from them. The attenuation correction based on the A_{H} - Z_{HH} empirical relation is also unstable and sensitive to the calibration error in Z_{HH} . The accuracy and stability of the attenuation correction scheme is vastly improved when dual-polarimetric observations are used (Bringi et al, 1990). Considering limited antenna aperture area on C-130 fuselage, deeper penetration





of C-band signals into squall lines and rainbands due to lower attenuation at C-band than at X-band, and the availability of more stable and accurate attenuation correction scheme using polarimetric radar measurements, C-band was selected as operating wavelength for the APAR.

224 4. Polarimetric measurement configurations

225 Weather radars make polarimetric measurements in two distinct transmit/receive modes: (i) alternate 226 transmit and simultaneously receive (ATSR), and (ii) simultaneously transmit and simultaneously receive 227 (STSR). Any cross-coupling between horizontally (H) and vertically (V) transmitted/receive waves in the 228 hardware and/or propagation medium due to canted hydrometeors would bias differential reflectivity (Z_{DR}) 229 measurement. For example, in STSR mode better than 44 dB isolation between H and V channels is 230 required to insure that bias in Z_{DR} is less than 0.2 dB (Wang and Chandrasekar 2006). In the ATSR mode 231 better than 20 dB isolation between H and V channels is sufficient for limiting bias in $Z_{DR} < 0.2$ dB. Also, 232 in the STSR mode, cross-polarization measurement is not feasible as radar transmits in both horizontal and 233 vertical polarizations simultaneously. As the most desired measurement in a polarization configuration is 234 Z_{DR} , the alternate transmit mode is a preferred one for the proposed APAR. 235 APAR system can be designed to operate in one of two alternating transmit modes:

• ATSR -- requires two receive channels

• ATAR (alternate transmit and alternate receive) -- requires one receive channel.

In the alternating transmit mode, both cross and co-pol measurements i.e. full scattering matrix is available. The ATSR mode requires twice the amount of time as the STSR mode for acquiring measurements as it alternates between H and V transmit polarizations.

The primary polarimetric capability is to compute co-polar parameters e.g. Z_{DR} , K_{DP} , and ρ_{hv} . To achieve this objective, horizontal and vertical antenna patterns (main beam) must be in excellent agreement spatially. This not only applies to broadside but as the beam is scanned in azimuth and elevation. A secondary polarimetric capability is to produce quality cross-polar measurements, like LDR. Two factors





govern the ability to make accurate cross-polar measurements. First, there needs to be sufficient signal in the co-polar channel; SNR in the co-polar channel must be greater than or equal to the LDR or the hydrometeors. Second, isolation between the co-polar and cross-polar channels must be several dB greater than the LDR lower limit. In practice this isolation is relatively easy to achieve in the T/R module and downstream radar hardware. So, antenna performance is usually the limiting factor. The antenna's integrated cross-polar ratio ICPR (two-way) is the defining property. ICPR is defined as

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$$ICPR = 10\log_{10}\left[\frac{\int_{0}^{2\pi\pi/2} \int_{0}^{\pi/2} \left|f_{hh}f_{vh} + f_{hv}f_{vv}\right|^{2}\sin(\theta)d\theta d\phi}{\int_{0}^{2\pi\pi/2} \int_{0}^{\pi/2} \left|f_{hh}^{2} + f_{hv}^{2}\right|^{2}\sin(\theta)d\theta d\phi}\right]dB, \quad (1)$$

Where: f_{hh} and f_{vv} are the co-polar antenna patterns (amplitude and phase), f_{vh} and f_{hv} are the crosspolar antenna patterns (amplitude and phase). The first index in the subscript denotes the transmit polarization, the second index denotes the receive polarization. ϕ is azimuth angle in radians and θ is the elevation angle in radians.

257 For estimating intrinsic LDR of -27 dB within 1 dB error, transmit/receive isolation i.e. ICPR must be < -33 dB (Bringi and Chandrasekar, 2001). This applies not only to broadside, but also as the beam is scanned 258 259 in azimuth and elevation. In the case of an electronically scanned aperture array (ESA) when a beam is 260 steered electronically away from the boresight, the transmitted field is biased as a function of scan angle and 261 cross coupling between dual polarization sources occur (Zhang et al. 2009). In order to limit the effect of 262 differential gain and beam pattern on polarimetric measurements, co-polarization measurements will be 263 collected only up to 20 degrees from broadside. A detailed discussion regarding gain changes and cross 264 coupling is presented in Section 10.

Another key aspect of phased array radars is the ability to form multiple, simultaneous beams upon reception using digital beamforming techniques. Although this capability would not routinely be part of the





267	APAR mission, in certain high impact weather cases it would be extremely useful. Specifically, it would be
268	desirable to have the ability to spoil the transmit beam to allow for a 20° fan beam in elevation, steerable in
269	azimuth and receive a separate RF signal(s) for each row of the array. A brief description of various PAR
270	architectures with regard to digital and analog beamforming is presented in Section 8.

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272 **5. Agile beam scanning**

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274 The measurement accuracy of reflectivity and Doppler radial velocity is a function of time-to-275 independence (T_p), PRF and signal-to-noise ratio (SNR) (Doviak and Zrnic, 1993). Time-to-independence 276 determines the interval between two radar measurements that are statistically independent. It is a function of 277 transmit frequency and spectrum width (Bringi and Chandrasekar 2001). At C-band, $T_{\rm p}$ is 6.2 ms for a 278 Doppler spectrum width of 1 m/s. The beam multiplexing technique allows for a reduction in the dwell time 279 needed to acquire a sufficient number of independent samples. Averaging the signals of independent 280 samples reduces fluctuation in radar estimates of wind and reflectivity. Figure 4 shows contiguous pulses 281 and beam multiplexing pairs. The minimum time to independence interval between beam multiplexing 282 pairs is $T_{\rm D}$. The maximum allowable dwell time on a radar sampling volume depends on the advection time 283 of the storm. For maximum allowable dwell time of 2.33 s, weather advection of 28.8 m s⁻¹ (60 miles per 284 hour), more than 75 % of weather echo remains in sampling volumes at ranges > 14 km (Curtis, 2009).







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Figure 4. Illustration of contiguous and beam multiplexing pairs (Curtis, 2009).

288 Typical scan sequence in less than 2 s time interval would consist of three types of scans: (i) dual-289 Doppler, (ii) dual-Doppler and dual-polarization, and (iii) surveillance. Dual-Doppler (DD) mode will be 290 the primary mode of operation. In this mode, each of the four AESAs will generate a single "pencil" beam that will be scanned in azimuth and elevation. Scanning in azimuth will be between two fixed angles, one 291 fore, and one aft. The fore and aft azimuth angles will be separated by 35° (Vivekanandan et al. 2014) Each 292 293 of the twenty-five elevation angles will be covered with about 25 independent pulse pair radar 294 measurements at each beam position in 1 s. Dual-polarimetric measurements will be collected by 295 transmitting alternatively in horizontal and vertical polarizations in the fore direction in dual-polarization 296 and dual-Doppler (DPDD) scan mode. Polarimetric measurements will be collected only over 10 elevation 297 angles in 0.4 s. A secondary surveillance scan mode of duration 0.4 s will be interleaved every 30 to 60 s 298 periodically with the primary mode to produce a "composite" PPI scan which incorporates data from the





- three aft mounted AESAs and the weather avoidance radar located in the nose of the aircraft. In surveillance
- 300 mode, the elevation angle is held fixed at 0^0 and the beam is scanned in azimuth.
- 301 **6. APAR antenna aperture**
- 302 This section considers three aperture shapes for the Active Element Scanning Array (AESA) and assesses
- their suitability to APAR. Individual element radiation pattern and array factor determine the radiation
- 304 pattern of an AESA. Radiation elements could be dipoles, waveguide slots, or microstrip antenna patch.
- 305 These elements arranged in a plane in rectangular, square, circular or elliptic shape defines a planar array.







307 Figure 5. Elliptical, circular and square phased array antenna apertures.

308 The radiation pattern of a planar array is characterized by directivity, 3 dB beam width, side lobe 309 level, and cross-pol isolation. Radiation characteristics of rectangular, square and circular planar arrays with 310 crossed dipole element as a radiating element are studied and are depicted in Figure 5. Available maximum 311 fuselage area for AESA on a C-130 aircraft is elliptical-shape with major and minor axes of 1.93 m and 1.78 312 m respectively. This area could hold about 3562 radiating elements at half wavelength spacing. At C-band 313 center frequency of 5.45 GHz, the half wavelength spacing is 2.75 cm. Figure 6 shows uniformly weighted 314 array factor patterns for the elliptical, circular and square apertures. The maximum circular area in that 315 elliptic area accommodates 3280 radiating elements whereas a square aperture will hold only 2304 316 elements. For matched beamwidths in azimuth and elevation, a square or circular aperture is preferred but 317 square aperture would transmit 30 % less power than a circular aperture. Also, for a uniform illumination, 318 the circular aperture has 4 dB lower peak sidelobes than a square aperture.







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321 Figure 7 shows transmit and receive patterns of the circular aperture for three different taperings: (i) 322 uniform tapering for both transmit and receive, (ii) uniform transmit and 30 dB Taylor taper for the receive 323 and (iii) 15 dB and 30 dB Taylor tapering of transmit and receive aperture. Since the two-way antenna 324 pattern is a product of transmit and receive patterns. In transmit and receive modes when all of the planar 325 elements are tapered uniformly maximum sensitivity is realized but peak side lobe (PSL) and integrated 326 sidelobes (ISL) are the highest. Lower PSL and ISL are desired for minimizing spatially disparate targets 327 from biasing the measurement of interest. When transmit and receive apertures are tapered PSL and ISL are 328 minimum but sensitivity is lower by 1.6 dB when compared to uniform tapering of aperture in transmit and 329 receive modes. As expected the receiver pattern has a lower sidelobe but its main beam is broader.





Typically, in the receive mode non-uniform weighting across planar array is applied for realizing a lower sidelobe and the uniform transmit weighting is applied for realizing maximum power aperture factor, the resultant two-way pattern is expected to have a desired lower sidelobe. When receive aperture is tapered 30 dB and maximum power is transmitted by uniform weighting of the aperture, relative sensitivity is decreased only slightly and also PSL and ISL were lowered by 8 and 10 dB respectively.

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Figure 7. Transmit and receive patterns of the circular aperture for three different tapering are shown: (i) uniform tapering for both transmit and receive, (ii) uniform transmit and 30 dB Taylor taper for the receive and (iii) 15 dB and 30 dB Taylor tapering of transmit and receive aperture

7. Radar sensitivity and prime power





342 The limited supply of aircraft power (prime power) available for the phased array system results in very 343 real limitations on radar performance. Prime power for the four AESAs on the NSF/NCAR C-130 is limited 344 to 30 KW. Since the primary mission of airborne weather radar is to sample hydrometeors, scan coverage of 345 the top and bottom AESAs can be limited to 25 km and ground level, respectively. This effectively reduces 346 the combined duty cycle on receive for all four AESAs to \sim 60%, thus conserving prime power. Figure 8a 347 provides the radar sensitivity for the three apertures considered, for both ATAR and ATSR TR module 348 architectures, and for both 4W and 6W peak power HPAs. In several cases, transmit duty cycle was reduced 349 to meet prime power constraints, resulting in reduced sensitivity at 10 km.

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- Figure 8a. Sensitivity comparison for three apertures, using two T/R module architectures (ATAR, ATSR)
- 355 and two different HPAs (4W, 6W)





356 The prime power is estimated based on the preliminary characterization of a "brick" T/R module 357 that has been developed in-house (Salazar et al. 2015). The microwave circuit technology employed in this 358 T/R module is GaAs. The T/R module is comprised of discrete MMICs arranged in an ATAR architecture. 359 The following conclusions can be drawn from this exercise: (i) the square aperture provides the worst 360 sensitivity, (ii) the use of an ATSR architecture results in ~ 3 dB loss in sensitivity for a given aperture 361 choice, and (iii) the sensitivity of the circular aperture is < 1 dB worse than that of an elliptical aperture operated with the same T/R module architecture and HPA. Figure 8b shows sensitivity of the APAR as a 362 363 function of range. Long pulse is used enhancing sensitivity via pulse compression. It is noteworthy to point 364 out desired sensitivity of -11 dBZ at 10 km is achievable either in ATSR or ATAR modes in any of the 365 three aperture configurations with 6 or 4 W high power amplifiers (HPA).



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Figure 8b. Sensitivity of APAR as a function of range.

368 8. Radar architecture





369 Phased array radar elements could be distinctly configured in three types of architecture: (i) analog, 370 (ii) hybrid, i.e. combination of analog and digital, and (iii) digital. In an analog array RF phase shifters and 371 attenuators are used for steering the beam (Herd and Convey, 2016). The performance of RF components 372 are sensitive to temperature and their precision is limited by the quantization or number of bits that are used 373 to represent phase and attenuation. Beamforming is achieved by summing signals from individual receive 374 elements by an analog combiner. The front-end analog combiner fixes antenna beamwidth and sidelobes 375 and they can not be modified. In a fully digital array, RF phase shifters and attenuators are replaced by 376 complex multiplication using digital electronics. As digital electronics based complex multiplication is 377 relatively immune to quantization effects, precise control of phase and amplitude is realized. Each antenna 378 element is coupled to individual receiver-exciter. In a digital array, an analog to digital converter (ADC) is 379 placed in front-end of the every RF element. The received signals are digitized at the element level and they 380 are converted to complex baseband (I/Q) samples.

381 Data rates of fully digital array radar are many orders of magnitude larger than a comparable analog 382 array. An hybrid architecture using a combination of analog and digital beamformer could reduce the date 383 rate to a manageable level for real-time data processing.

Table 2. Comparison between analog and digital architectures

Attributes	Analog	Digital
Hardware	RF phase shifters and attenuators are analog. Analog phase are expensive and performance varies with temperature and between T/R modules	Phase shifters and attenuators are implemented by digital operations: no quantization effects enable precise phase and amplitude control





Antenna beamwidth	Fixed for specified transmit and	Fully re-configurable in post-processing;
sidelobe and gain	receive modes; depends on aperture	multiple beams could be formed
	weighting	
Receiver	Single receiver chain; simple and	Multiple distributed receivers.
	inexpensive	Dynamic range Improvement: 10log10(N).
		Increased complexity and expense
Digitization of the	Weighted sum of N radiating	Complete access to N antenna element
received signals	element received signal	signals
Calibration and data	Performance of RF Phase shifters	Linear and robust performance
quality	and attenuators are sensitive to	
	temperature	
Data rate	100MB/s	4GB/s

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386 Examples of analog, hybrid and digital array architectures are illustrated in Figure 9. Analog 387 beamformer using RF phase shifters and attenuators in ATAR configuration is shown Figure 9a. Beams 388 could be steered to contiguous azimuth and elevation angles. The number of variable phase shifters and 389 attenuators could be reduced by the square root of number of the number of elements by using fixed phase 390 shifters as shown in Figure 9b. For steering the beam between two discrete beam positions separated by 35° , 391 fixed phase shifters are used. As alluded to earlier, 2-D winds will be measured using the fore and aft beams separated by 35⁰. The antenna beam could be contiguously steered in elevation. The surveillance scan could 392 393 not be realized in this configuration as the beams could not be steered contiguously in azimuth. Figure 9c 394 shows a hybrid architecture with no variable phase shifters and attenuators. This configuration allows digital 395 beamforming in elevation but does not allow scanning in the azimuthal direction. Table 2 summarizes 396 salient features of analog and digital architectures. Based on the technical specification described in the 397 Table 1, the data rate for analog configuration is 100MB/s whereas the digital array radar in 1-D hybrid 398 architecture with no variable phase shifters and attenuators would generate 4 GB/s data rate. These data





rates are many orders of magnitude larger than a dual-channel polarimetric mechanically scanning radar as
S-Pol (Lutz et al. 1997) that generates 8 MB/s. Since the configuration shown in Figures 9b and c uses
fixed phase shifters for steering the beam in two fixed azimuth directions, they are not capable of scanning
in azimuth.

103 Not having continuous beam steering capability in azimuth limits acquisition of a composite PPI 104 'surveillance' scan from either side of the fuselage radars, radar located on upper portion of the tail ramp 105 radars in combination with C-130 weather avoidance radar (Vivekanandan et al. 2014). This "surveillance" 106 mode is essential to provide safety in single aircraft missions and will also aid in mission flight planning 107 while in the air. The PAR on top of the fuselage could be operated as an end-fire array for the acquisition of 108 surveillance scan. It should be noted not having a broadside array on top of the fuselage limits dual-109 Doppler scans that are suitable for retrieving vertical winds of over the head storms. End-fire phased array 110 radar has lower gain compared to a broadside array. Re-directing the usage of the radar on top of the **1**11 fuselage would diminish redundancy in APAR's ability in collecting volume scans. Instead of a PAR, a **1**12 mechanically scanning radar as in the NOAA P-3 could provide a dedicated surveillance scan feature. **413** Potential usefulness of end-fire PAR and mechanically scanning radar for surveillance scan application will 114 be investigated in the future. Additional studies are necessary for understanding pros and cons of operating **115** the radar on top of the fuselage in broadside and end-fire configurations.







2-D Analog Beamformer (ATAR)

116









Figure 9. APAR architectures for ATAR configuration: (a) 2-D analog, (b) 1-D analog, and (c) 1-D hybrid.

121

9. Calibration of reflectivity

Weather radar applications place high demands on phased array calibration. It is critical to know accurately the gain, beamwidth and pointing angle of the array over a variety of operating conditions. Calibration is divided into three phases: (i) characterization, (ii) field calibration and (iii) end-to-end calibration using an external target such as the Sun or a rain medium. Characterization is performed under controlled conditions in an anechoic chamber. It is done prior to the arrays being placed in service or after





any major component failure/repair. Field calibration is done in-situ on the aircraft. It is performed after the arrays are installed on the aircraft prior to a field project or during a field project after a repair has been made. Field calibration must be able to compensate for component drift as well as any performance variations due to the departure of the installation environment from that of the anechoic chamber. The equipment, if any, required for field project calibration must be compact and transportable. The field calibration of a single array must be able to be completed within 10 hours, including any setup or teardown of equipment.

435

136 Reflectivity can be calibrated using a known signal source such as transmit power from a horn antenna in 137 the far zone or the solar radiation. When a known external signal source is used, the radar does not transmit, 138 and only the receiver system is calibrated. In the case of solar calibration [reference], the receiver should be 139 sensitive enough to detect the low signal power (100 dBm). Also, the main lobe beam width should be less 140 than 0.5° for satisfying the beam-filled condition. In the transmit and receive mode, a highly reflective test 141 sphere suspended from a tethered balloon can be used as a known reference target for calibration. But the 142 sphere being a point target does not fill the radar beam; as a result, only the on-axis gain of the beam is 143 measured. Self-consistency among reflectivity, differential reflectivity, and propagation phase can be used 144 for calibrating the radar system (Vivekanandan et al. 2003). One of the advantages of using power and 145 phase measurements from a single radar is the elimination of sampling volume differences among the 146 measurements. At C-band, reflectivity and differential reflectivity must be corrected for attenuation and 147 differential attenuation prior to calibration based on self-consistency among radar measurements. 148 Attenuation and differential attenuation could be obtained from propagation measurement or a variational 149 method that use both differential reflectivity and propagation phase measurements (Chang et al. 2014)

450

10. Removal of bias in polarimetric radar measurements





In the case of a mechanically steered antenna, the horizontally and vertically polarized beams preserve intrinsic source polarimetric response independent of the beam direction. Since orthogonally between H and V polarization is desired for estimating cloud microphysical measurements of hydrometeors, radiation fields in dual-polarization radar are transmitted orthogonal to each other. In an ESA when a beam is steered electronically away from the boresight, the transmitted field is biased as a function of scan angle and cross coupling between dual polarization sources occur (Zhang et al. 2009). Figure 10 illustrates spherical coordinate system of planar array and hydrometeor scattering location at a range r.

159



160

Figure 10. The coordinate system for electric fields from a pair of microstrip radiating elements. \overline{M}^{h} is the magnetic current density of a horizontally polarized radiating element. \overline{M}^{v} is the magnetic current density of a vertically polarized radiating element Hydrometeors are located at range r. Unit vectors a_{r} , a_{θ} and a_{ϕ} form a local orthogonal system at r (Lei et al, 2015).

The relations among transmit polarization field vectors at the planar aperture, incident field vectors on a hydrometeor, and received fields can be described using the following linear transformation. The transmitted electric fields, $\vec{E}_{th}^{(p)}$ and $\vec{E}_{tv}^{(p)}$ generated by radiation sources M_h and M_v and projected onto the local H and V directions, Geosci. Instrum. Method. Data Syst. Discuss., https://doi.org/10.5194/gi-2017-45 Manuscript under review for journal Geosci. Instrum. Method. Data Syst. Discussion started: 6 September 2017





469
$$\vec{E}_{i} = \begin{bmatrix} E_{i\phi} \\ -E_{i\theta} \end{bmatrix} = \mathbf{P} \begin{bmatrix} E^{(p)} \\ \mathbf{h} \\ E^{(p)} \\ \mathbf{t} \end{bmatrix} = \mathbf{P} \vec{E}_{i}$$
(2)

170 where the projection matrix \mathbf{P} is (Lei et al. 2015)

471
$$\mathbf{P} = \begin{bmatrix} p_{11} & p_{12} \\ p_{21} & p_{22} \end{bmatrix} = \begin{bmatrix} E_{\phi}^{(h)} & E_{\phi}^{(v)} \\ -E_{\theta}^{(h)} & -E_{\theta}^{(v)} \end{bmatrix}$$
(3)

172

173 where the superscripts define H and V antenna ports. The H antenna port primarily transmits 174 horizontally polarized fields and V antenna port primarily transmits vertically polarized fields, and all the 175 electric fields on the right side of above equation are normalized by their respective broadside electric fields

176 (e.g.,
$$E_{\phi}^{(h)}(\theta, \phi)$$
 or $E_{\theta}^{(h)}(\theta, \phi)$ is normalized by $E^{(h)}(\frac{\pi}{2}, 0)$

177 The projection matrix for microstrip patch antenna is (Lei et al. 2015)

$$\mathbf{P} = \begin{bmatrix} \sin\theta \cdot g^{(h)}(\theta,\phi) & \cos\theta\sin\phi \cdot g^{(v)}(\theta,\phi) \\ 0 & \cos\phi \cdot g^{(v)}(\theta,\phi) \end{bmatrix}$$
(4)

where radiation fields of microwave patch antenna $E_{\phi}^{(\nu)}(\theta,\phi)$, $E_{\phi}^{(h)}(\theta,\phi)$, $E_{\theta}^{(\nu)}(\theta,\phi)$, and $E_{\theta}^{(h)}(\theta,\phi)$ are 179 related to $g^{(h)}(\theta,\phi)$, and $g^{(v)}(\theta,\phi)$ (Balanis 1997). The projection matrix, **P**, couples oblique fields E^{h} and 180 181 E^{ν} to the local H and V coordinates at the scatterer and at the ESA aperture. The horizontally polarized 182 field is a function of $\sin(\theta)$ and vertically polarized field is a sum of intrinsic vertically polarized weighted 183 by $\cos(\phi)$ and a leakage term from cross-polarization (V) weighted by $\cos(\theta)\sin(\phi)$. The cross-coupling 184 term p_{21} could be neglected as cross-polarization isolation for a microstrip antenna is typically lower than 20 185 dB. The backscattered fields, E_r in the local H and V directions is a product between E_t and backscatter 186 matrix S. The received fields E_r at the electronic scanning antenna (ESA) aperture in alignment with H and 187 V channels is a product of P_t and E_r . The backscattering matrix is





188

190

where S' the intrinsic backscatter matrix is

 $\mathbf{S}' = \begin{bmatrix} \dot{s}_{hh} & \dot{s}_{hv} \\ \dot{s}_{vh} & \dot{s}_{vv} \end{bmatrix}$ 191

 $\mathbf{S}^{(p)} \equiv \mathbf{P}^T \mathbf{S'P}$



In an ATSR mode alternately transmitted H and V pulses are used for retrieving radial velocity and 193 polarimetric radar estimates. The above matrix is rewritten to take into account of the Doppler phase shift 194 between H and V pulse period as

 $= \begin{bmatrix} p_{11}^{2}s'_{hh} + p_{21}^{2}s'_{vv} + p_{11}p_{21}(s'_{vh} + s'_{hv}) & p_{11}p_{12}s'_{hh} + p_{21}p_{22}s'_{vv} + p_{11}p_{22}s'_{hv} + p_{12}p_{21}s'_{vh} \\ p_{11}p_{12}s'_{hh} + p_{21}p_{22}s'_{vv} + p_{11}p_{22}s'_{vh} + p_{12}p_{21}s'_{hv} & p_{12}^{2}s'_{hh} + p_{22}^{2}s'_{vv} + p_{12}p_{22}(s'_{vh} + s'_{hv}) \end{bmatrix}$ (5)

495
$$\mathbf{S}^{(p)} = \begin{bmatrix} s_{hh}^{(p)}(2i) & s_{hv}^{(p)}(2i+1) \\ s_{vh}^{(p)}(2i) & s_{vv}^{(p)}(2i+1) \end{bmatrix} = \begin{bmatrix} s_{hh}^{(p)}(2i) & e^{-j2k_0\hat{v}T_s} \cdot s_{hv}^{(p)}(2i) \\ s_{vh}^{(p)}(2i) & e^{-j2k_0\hat{v}T_s} \cdot s_{vv}^{(p)}(2i) \end{bmatrix}$$
(6)

196 Since the projection matrix term p_{21} is zero, the individual elements of the received backscatter at 197 the ESA are

 $s_{\rm hb}^{(p)}(2i) = p_{11}^2 s_{\rm hb}'(2i)$ (7a) 198

499
$$s_{vh}^{(p)}(2i) = p_{11}p_{12}s_{hh}'(2i) + p_{11}p_{22}s_{vh}'(2i)$$
 (7b)

 $s_{\rm hv}^{(p)}(2i+1) = p_{11}p_{12}s_{\rm hh}'(2i+1) + p_{11}p_{22}s_{\rm hv}'(2i+1) \quad (7c)$

501
$$s_{vv}^{(p)}(2i+1) = p_{12}^2 s_{hh}'(2i+1) + p_{22}^2 s_{vv}'(2i+1) + p_{12} p_{22}(s_{vh}'(2i+1) + s_{hv}'(2i+1))$$
 (7d)

502 In the ATSR mode, the above is a set of linear equations (Zrnic et al. 2011). The intrinsic scattering matrix elements are estimated as projection matrix P and , backscatter measurements S^P are known. Element 503 at 2i+1 time interval is related to the element at 2i by the Doppler phase shift $e^{j2k_0\hat{v}T_s}$. The Doppler shift and 504 505 propagation phase are estimated from cross-correlation between consecutive estimates of co-polarization 506 scattering amplitudes. Unbiased estimates of polarimetric observables require calibrated transmit





amplitudes. Thus pulse-pair methods can be used for estimating unbiased Doppler and polarimetric
 measurements provided transmit amplitudes are known.

Once the backscatter amplitudes are estimated using the above equations, all of the polarimetric variables could be derived (Zrnic et al , 2011). The bias in polarimetric measurements is estimated for radar resolution volume populated with spherical hydrometeor with a baseline ρ_{hv} of 0.9. The estimated bias depends on radiation pattern of the radiating element. The effect of the array factor is not considered in this study.

514 The bias is estimated for stacked patch microstrip antenna radiation element described in Lei et al 515 (2015). Figure 11 shows biases in Z_{DR} , Φ_{DP} , ρ_{hv} and LDR are symmetrical with respect to changes in azimuth and elevation angles. For 0^0 elevation, as the azimuth angle increases, Z_{DR} is positively biased. For 516 517 both positive and negative elevation angles Z_{DR} is biased negatively and it varies with the azimuth angle. For azimuth and elevation angles $< 20^{\circ}$, the bias is < 0.25 dB. Φ_{DP} and ρ_{hv} exhibit no bias for 0° elevation 518 and the bias is less than 1° for elevation and azimuth angles $< 20^{\circ}$. Bias in ρ_{hv} is less than 1% for 519 elevation angle $< 20^{\circ}$. Bias in LDR significantly increases as the beam scanned away from bore sight. For 520 elevation angle $< 5^{\circ}$ and azimuth angle $< 20^{\circ}$ LDR is < -30 dB. 521

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533

Figure 11. Biases in Z_{DR} , Φ_{DP} , ρ_{HV} and LDR as a function azimuth and elevation scan angle of the radar beam.

- 536
- 537 **11. Summary**

APAR with dual-polarimetric and dual-Doppler capability allows concurrent estimates of microphysical
 (e.g., precipitation types and sizes, quantitative precipitation estimation) and 3-D winds in a precipitation





540 system. At present, no other instrument other than an airborne polarimetric Doppler phased array radar sys-541 tem has the potential to estimate 3-D winds and microphysics concurrently. Both ATAR and ATSR 542 polarimetric configurations require same amount of time for acquiring Z_{DR} , Φ_{DP} , and ρ_{HV} . Additional 543 receive channel in ATSR enables measurement of LDR at a faster pace but the ATSR configuration 544 requires twice the amount of receiver elements compared to the ATAR configuration. Multiple AESA 545 radars on the C-130 fuselage enhance spatial and temporal resolutions of measurements. Solid state, 546 compact T/R elements based on GaAs or GaN is the key enabling technology; only demonstrated hardware 547 and software sub-systems will be included in the design of the AESA.

Agile beam steering using the e-scan feature enables the collection of more independent samples in a specified time interval when compared to a mechanically scanning radar. Since an airborne radar has a limited amount of time to collect measurements over a specified sample volume, agile beam steering reduces uncertainty in radar measurements.

552 For the maximum available area for AESA on a C-130 aircraft, elliptical and circular apertures produce 553 almost same directivity, but the circular planar array is more desirable. For uniform illumination, the 554 circular aperture has 3 dB lower peak sidelobe than a rectangular or square aperture. Unlike elliptic, square, 555 or rectangular planar arrays, distortions in the array pattern of a circular array due to mutual coupling effect 556 are same for each element and this makes it easier to deal with the mutual coupling effect. With respect to 557 sensitivity, the elliptical aperture with 6W HPA offers ~1dB better sensitivity than the circular aperture and 558 \sim 3dB better than the square aperture. Circular aperture offers matched beams in horizontal and vertical 559 polarization transmission.

560

Phased array radar architecture is evolving toward a combination of analog and digital (hybrid) and fully
 digital architecture. From the perspective of robust performance, improved data quality, and adaptive beam





⁵⁶³ forming, digital array architecture is preferred. Since digital array radar produces higher data rates, practical

- requirement for processing high data rates needs to be evaluated.
- 565 Methodologies for estimating polarimetric observables are summarized based on the earlier published

results. Biases in Z_{DR} , Φ_{DP} , ρ_{hv} and LDR due to cross coupling between dual polarization sources are 566 567 symmetrical with respect to changes in azimuth and elevation angles and they are with in acceptable range for microphysical studies for azimuth and elevation angles $< 20^{\circ}$. In an alternate transmit mode, the 568 569 received voltages are a set of linear equations and can be solved either by pulse-to-pulse adjustment or 570 powers and correlations of received voltages. Radial wind and polarimetric observables can be estimated 571 from correlations of received voltages. Self-consistency among reflectivity, differential reflectivity, and 572 propagation phase will be used for absolute calibration of reflectivity. This would require unbiased 573 differential reflectivity and attenuation corrected reflectivity.

574

575 Appendix: List of Acronyms

- 576 AESA Active Electronically Scanned Array
- 577 APAR Airborne Phased Array Radar
- 578 ATAR Alternate Transmit and Alternate Receive
- 579 ATSR Alternate Transmit and Simultaneous Receive
- 580 ATT Attenuator
- 581 C-130 Four-engine turboprop military transport aircraft
- 582 dBZ Radar reflectivity factor
- 583 ELDORA Electra Doppler radar
- 584 Electra Turboprop airliner
- 585 EOL Earth Observing Laboratory
- 586 e-scan Electronic scanning





587	FPGA	Field Programmable Gated Array
588	Н	Horizontal
589	HPA	High Power Amplifier
590	ICPR	Integrated Cross Polar Ratio
591	K _{DP}	Specific Propagation phase
592	LNA	Low Noise Amplifier
593	LDR	Linear Depolarization Ratio
594	LRU	Line Replaceable Unit
595	MPA	Medium Power Amplifier
596	MMIC	Monolithic Microwave Integrated Circuits
597	MODIS	Moderate Resolution Imaging Spectoradiometer
598	MSL	Mean Sea Level
599	NCAR	National Center for Atmospheric Research
500	NOAA	National Oceanic and Atmospheric Administration
501	NOAA P3	
502	NRL P3	
503	NSF	National Science Foundation
504	NWRT	National Weather Radar Testbed
505	PAR	Phased Array Radar
506	PPI	Plan Position Indicator
507	PRF	Pulse Repetition Frequency
508	PS	Phase Shifter
509	P-3	Four-engine turboprop surveillance aircraft
510	QC	Quality Control





511	QPE	Quantitative Precipitation Estimation
512	QPF	Quantitative Precipitation Forecast
513	RF	Radio Frequency
514	RHI	Range Height Indicator
515	Rx	Receiver
516	SNR	Signal-to-Noise Ratio
517	Solo	Software for radar translation, visualization, editing and interpolation
518	STSR	Simultaneously Transmit and Simultaneously Receive
519	T _D	Time-to-independence
520	T/R	Transmit/Receive
521	Tx	Transmitter
522	V	Vertical
523	Ζ	Reflectivity
524	Z _{DR}	Differential reflectivity
525	λ_{o}	The transmit wavelength
526	$\Phi_{ m DP}$	Propagation phase
527	θ_0	Beamwidth along boresight of the antenna

528

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