## A steerable system of RECoverable Autonomous Sonde (RECAS) for accessing and studying subglacial lakes

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#### 9 Abstract

10 The study of subglacial lakes requires clean access and sampling technologies. One of the most promising alternatives is the newly developed thermal drill RECoverable Autonomous Sonde 11 (RECAS), which allows downward and upward ice drilling and subglacial water sampling while 12 13 the subglacial lake remains isolated from the surface. The original sonde descends downward under the force of gravity, and the borehole trajectory cannot be controlled. However, in certain 14 cases, the sonde would preferably be able to drill at specific angles and directions, enabling it to 15 follow a desired trajectory (e.g., maintaining verticality within the desired range) or bypass 16 obstacles in the ice (e.g., stones and other inclusions). The general principle for the steering 17 18 RECAS is to adjust the voltage for the electric thermal head heaters, which provides an opportunity to control the heat distribution on the thermal head surface, thereby altering borehole trajectory 19 during drilling. In this paper, the general principles of steering RECAS are described, and 20 21 experimental results on deviational ice drilling with a controllable electric thermal head are 22 discussed.

#### 23 Keywords

24 Ice drilling technology; Subglacial lakes; Clean access sampling; Thermal sonde; Thermal drill; Steerable system 25

#### 26 **1. Introduction**

27 It is now widely accepted that subglacial hydrological environments are similar to the water distribution found elsewhere on Earth's surface and comprise a vast network of lakes, rivers, and 28 29 streams located thousands of metres beneath ice caps, glaciers, and the Antarctic and Greenland ice sheets (Bowling et al., 2019; Siegert et al., 2012a; Ashmore et al., 2014). A subglacial lake is 30 considered to be any large body of liquid water existing below an ice mass. The water depth of 31 32 subglacial lakes varies from a few to several hundred metres (Wright et al., 2011). As of 2022, a total of 773 subglacial lakes have been identified, including 675 in Antarctica, 64 in Greenland, two beneath 33 the Devon Ice Cap, six beneath Iceland's ice caps, and 26 in valley glaciers (Livingstone et al., 2022). 34 35 The ice thickness above subglacial lakes may vary from several tens to thousands of metres.

Subglacial lakes provide unique information regarding paleoclimatic conditions, basal hydrology, 36 biogeochemical fluxes, and geomorphic activity. It is anticipated that subglacial lakes harbour relict 37 microbial species capable of thriving in complete darkness, low nutrient levels, high water pressures, and 38 isolation from the atmosphere (Skidmore, 2011). In-situ investigations should not contaminate these 39 40 subglacial aquatic systems. Currently, hot-water drilling systems are considered the cleanest method for accessing subglacial lakes. US teams successfully accessed the Whillans and Mercer subglacial lakes 41 on the coastal margin of West Antarctica in early 2013 and during the 2018-2019 season, demonstrating 42 the well-proven effectiveness of this technology (Priscu et al., 2021; Tulaczyk et al., 2014). 43

However, access technology using hot-water drilling systems has several significant limitations. 44 For instance, these systems necessitate complicated methods to filter and subject high-speed hot-water 45 46 flow (>150-200 L/min) to ultraviolet (UV) treatment at the surface. Additionally, they are extremely bulky and highly power-consuming. For example, all drilling and auxiliary equipment for exploring 47 of Subglacial Lake Ellsworth in West Antarctica weighed approximately 60 t, and the fuel supplies 48 weighed a further approximately 55 t (Siegert et al. 2012b). To simplify the drilling process and 49 decontamination of drilling tools, we propose accessing and studying subglacial lakes with a freezing-50 in electric thermal drill — the RECoverable Autonomous Sonde (RECAS) — capable of downward 51 and upward ice drilling and subglacial water sampling while ensuring that the subglacial lake remains 52 isolated from the surface (Talalay et al., 2014). RECAS is estimated to be 10-20 times less expensive 53 than penetration with a hot-water drilling system, and its installation and operation require only four 54

specialist staff members (Sun et al., 2023) vs. at least 10 staff required to run a deep hot-water drill (Siegert et al., 2012). The whole RECAS system including sonde, winch, generator, etc. (without fuel supply) weighs about 6.2 t. The sonde surface is thoroughly cleaned before deployment. Although the sonde might drag native microbes, which are embedded in ice, into subglacial targets at various depths as they melt, this occurs in a predictable manner (Schuler et al., 2018). Two concepts similar to RECAS have been proposed by Stone Aerospace, a US engineering company (Pereira et al., 2023; Stone et al., 2018), and Aachen University in Germany (Heinen et al., 2021).

The RECAS was successfully tested in East Antarctica during the 2021-2022 field season, 62 63 reaching the ice-sheet base at a depth of 200.3 m, sampling basal meltwater and measuring its pressure, temperature, pH, and conductivity before returning to the ice surface (Sun et al., 2023). To 64 expand the sonde's capabilities, we propose equipping it with a steering technique to control and 65 guide the drilling process. This allows drilling at specific angles, depths, and directions, enabling the 66 sonde to follow a desired trajectory (e.g., maintaining verticality within the desired range) or bypass 67 obstacles in the ice (e.g., stones and other inclusions). A similar system was implemented in the 68 IceMole probe (Dachwald et al., 2014). IceMole has an original steering system consisting of a 69 combined thermomechanical drill head and controlled lateral heating. However, the probe is 70 71 significantly shorter (1 or 2 m in length), has a square cross-section, and can only operate in dry boreholes (maximum drilling depth ~25 m), which differs greatly from the operating conditions of 72 73 RECAS. Due to the complexity of adapting the IceMole system to RECAS, we developed original 74 steering system. Herein, we describe the general principles of the steering RECAS and discuss the experimental results on deviational ice drilling with a controllable electric thermal head. 75

76

#### 2. Steering approaches of the RECAS

#### 77 2.1. General concept of the steering RECAS

The RECAS comprises four major systems: a heating system (consisting of an upper thermal head, a lower thermal head, and lateral heaters), an inner winch system, a scientific load platform, and a monitoring and control systems (Sun et al., 2024). The upper and lower thermal heads are identical except for the central hole of the cable in the top thermal head (Li et al., 2020). Thus, it can drill both downward and upward and move within the borehole using an inner cable-recoiling mechanism, similar to how a spider climbs on its silk line.

Two RECAS prototypes were developed: RECAS-200, with a 200-m-long cable inside, and RECAS-500, with a 500-m-long cable inside. The RECAS-500 design is shown in Fig. 1. The prototypes differed not only in their drilling ability but also in their sizes, power consumption, number of cartridges in the thermal head, etc. (Table 1). In both prototypes, all heaters are supplied simultaneously at the same voltage from a single source.

- 89 **Table 1**
- 90 General parameters of RECAS-200 and RECAS-500

Prototype	Diameter, mm	Total length, m	Total power, kW	Power of thermal head, kW	Num. of cartridges in thermal head
RECAS-200	160	7.9	8.8	5 nom.; 6 max.	16
RECAS-500	180	7.3	9.7	6.5 nom.; 9.5 max.	20

91 The fundamental principle behind an adjustable electric thermal head is to control the voltage 92 supplied to each pair of adjacent heaters. This enables control over the heat distribution on the 93 thermal head surface and allows borehole deviations during drilling. Furthermore, controlling the 94 heat distribution of the thermal head makes it possible to equalise the load on the heaters as needed. 95 Heating cartridges exhibit variations in their parameters owing to their technological tolerances. 96 Additionally, these parameters may change slightly during long-term use, and heating cartridges 97 can fail because of their long-term use or manufacturing defects.

The sonde is steered using data from an inclinometer installed inside it. The data from the 98 inclinometer are transmitted to a personal computer (PC), processed, and converted into pulse-99 width modulation (PWM) coefficients, which determine the PWM duty cycle for a specified 100 101 number of channels. In the subsequent tests using the RECAS-200 prototype, the PC will be replaced with a microcontroller mounted inside the sonde. The PMW coefficients are transmitted 102 103 from the computer to a PWM generator (Sup. 2) inside the sonde prototype, where an individual 104 PWM signal is generated for each channel. Each PWM signal is amplified using a power module 105 (Sup. 3) and supplied to the corresponding heater inside the thermal head. The PWM signal duty 106 cycle limits the heater power.

107 It was decided to implement three control modes: manual mode, where the heaters 108 configuration can be set manually; semi-automatic mode, where the configuration is "linked" to 109 the tilt angle of the sonde; and automatic alignment mode, where the system will "attempt" to align 110 the sonde to a vertical position and/or maintain the sonde vertically.





Fig. 1. General schematic of the RECAS-500 with a 500-m-long cable inside (all dimensions are in mm)

#### 113 2.2. RECAS positioning estimation

The following method was employed to convert the values received from the inclinometer into PWM coefficients for automatic alignment mode. The inclinometer transmitted deviation values in degrees along the *X* and *Y* axes. As shown in Fig. 2, the *X* and *Y* axes correspond to the inclinometer axes in the horizontal plane. Point *A* indicates a deviation, with X = 4.5 and Y = -3, for instance. Eight pairs of 16 heaters are shown schematically in the form of circles, designated as *H1-H8*.



119

**Fig. 2.** Schematic diagram of the heater pairs *H1-H8* relative location and inclinometer in the sonde prototype. (a) projection length l, (b) heater pair axis angle  $\alpha$ 

#### 122 First, the absolute inclination $\varphi$ is determined (Eq. 1). This value is required not only for

123 subsequent calculations but also for monitoring purposes.

$$\varphi = \sqrt{X^2 + Y^2} \tag{1}$$

125 Next, the projection length values  $l_n$  of  $\varphi$  on the axis of each heater pair are determined as

126 follows (indicated in red in Fig. 2a):

$$l_n = \varphi \cos\left(\arctan\left(\frac{Y}{X}\right) - \alpha_n\right)$$
(2)

127 where  $a_n$  is the heater pair axis angle relative to the *X* axis (Fig. 2b).

To obtain the required PWM coefficients, the  $l_n$  values must be converted to relative values in the range of 0-1. Additionally, it is necessary to be able to adjust the resulting coefficients. For this purpose, a logistic function (logistic curve) was used (Kyurkchiev et al., 2015). After slight adaptation to meet our conditions, the final equations take the following form:

$$K_{n} = \frac{1}{1 + \exp(-T(l_{n} + V))}; V = \frac{1}{T} \ln \frac{-y_{off}}{y_{off} - 1}$$
(3)

where  $K_n$  is the PWM coefficient for each heater pair, *V* is the intermediate coefficient, *T* is the correction coefficient (above zero), and  $y_{off}$  is the offset coefficient (0-1).

T and  $y_{off}$  are used to adjust the final values. The coefficient  $y_{off}$  limits the maximum average PWM coefficient value (i.e., with zero inclination and  $y_{off} = 0.5$ , all PWM coefficients will be 0.5). Meanwhile, the correction coefficient *T* affects the rising section length where the derivative is relatively large. The influence of *T* and  $y_{off}$  on the final results is illustrated by the example discussed next.

### 139 2.3. RECAS positioning calculation example

140 For the calculation example, random inclinometer values are taken as: X = 4.5; Y = -3. Then, 141 absolute inclination is

142 
$$\varphi = \sqrt{X^2 + Y^2} = \sqrt{4.5^2 + (-3)^2} = \sqrt{20.25 + 9} = \sqrt{29.25} = 5.41.$$

- 143 The  $\alpha$  values for eight heater pairs are presented in Table 2.
- 144 Table 2
- 145 The  $\alpha$  values for eight heater pairs

$\alpha_l$	$\alpha_2$	a3	α4	α5	$\alpha_6$	α <sub>7</sub>	$\alpha_8$
0°	45°	90°	135°	180°	225°	270°	315°

### 146 The projection length of $\varphi$ for the first heater pair is estimated as

147 
$$l_1 = \varphi \cos\left(\arctan\left(\frac{Y}{X}\right) - \alpha_1\right) = 5.41 \cdot \cos\left(\arctan\left(\frac{-3}{4.5}\right) - 0\right) = 5.41 \cdot \cos\left(-33.69\right) = 5.41 \cdot 0.83 = 4.5.$$

The calculation results for all eight projection length values are listed in Table 3 and shownin graph form in Fig. 3.

#### 150 Table 3

152 153

151 Projection length values



154 For this example, the following coefficients were selected: T = 1 and  $y_{off} = 0.8$ . Then, the



156 
$$V = \frac{1}{T} \ln \frac{-y_{off}}{y_{off} - 1} = \frac{1}{1} \cdot \ln \frac{-0.8}{0.8 - 1} = \ln 4 = 1.39$$

157 
$$K_1 = \frac{1}{1 + \exp(-T(l_1 + V))} = \frac{1}{1 + \exp(-1 \cdot (4.5 + 1.39))} = \frac{1}{1 + \exp(-5.89)} = \frac{1}{1 + 0.0028} = 0.997$$

- 159 **Table 4**
- 160 PWM coefficients

K1	K2	K3	K4	K5	K6	K7	K8
0,997	0,92	0,17	0,02	0,04	0,58	0,99	0,999

161 To illustrate how *T* and *y*<sub>off</sub> affect the final PWM coefficients, the calculation results with the

same deviation values but different T and  $y_{off}$  values are presented in Figs. 4 and 5.







166Fig. 5. PWM coefficients at constant X = 4.5; Y = -3; T = 1, and three different values of  $y_{off} = 0.2$ ;  $y_{off} = 0.5$ ;  $y_{off} = 0.8$ .167The coefficients T and  $y_{off}$  should be selected experimentally. Therefore, they do not need to168remain constant. They may depend on other parameters; for example,  $y_{off}$  may depend on the169absolute inclination. It is worth noting that when the correction coefficient T approaches zero, all170PWM coefficients tend towards the  $y_{off}$  value, which means that the heat distribution on the thermal171head surface approaches a uniform pattern.

#### 172 **3.** Passability of the RECAS

165

Before changing the borehole trajectory direction, determining the passability of the sonde in the drilled borehole is essential. Owing to its length exceeding 7 m, the RECAS has a high likelihood of becoming stuck in the borehole, even with relatively small deviations. The main parameter affecting sonde passability in a curved borehole is the deviation intensity. To characterise the borehole deviation intensity at a specific interval along its axis, the relative zenith angle values were used, considering the interval between their measurement points. Therefore, the zenith deviation intensity was determined as follows (Zvarygin, 2010; Shamshev et al., 1983):

$$i_{\theta} = \frac{\Delta \theta}{\Delta L} \tag{4}$$

180 where  $\Delta \theta$  is the relative zenith angle in degrees and  $\Delta L$  is the borehole axis interval length.

181 If the deviation intensity at some borehole interval remains constant ( $i_{\theta} = \text{const}$ ), it means 182 that the borehole is curved along a circular arc at a certain interval. The borehole radius of 183 curvature *R* depends on the deviation intensity, as follows (Zvarygin, 2010) (Fig. 6):

$$R = \frac{57.3}{i_{\theta}} \tag{5}$$



Fig. 6. Schematic of the stuck sonde in a curved borehole.

The passability of a sonde in a borehole interval with known diameter and radius of curvaturecan be determined as follows:

$$l_{s} \leq \sqrt{8R(D_{b} - D_{s}) - 4(D_{s} - D_{b})^{2}}$$

$$\tag{6}$$

where  $l_s$  is the length of the cylindrical part of the sonde (the thermal head length is not included), D<sub>s</sub> is the sonde diameter, and  $D_b$  is the borehole diameter.

As the borehole radius of curvature is considerably larger than the gap between the sondeand borehole diameters, Eq. 6 can be simplified as follows (Shamshev et al., 1983):

$$l_s \le \sqrt{8R(D_b - D_s)} \tag{7}$$

Based on RECAS field tests, the difference between the borehole and sonde diameters are 193 10-20 mm. This clearance mainly depends on the rate of penetration (ROP), and additional 194 laboratory tests are required to establish a more precise relationship. Considering that the RECAS length is approximately 7 m, the range of radii of curvature ensuring RECAS passibility is in therange of 300-600 m.

197 Therefore, it is not sufficient to simply monitor the borehole inclination to avoid the RECAS 198 from being stuck in the borehole. Instead, it is necessary to continuously estimate the deviation 199 intensity, the borehole radius of curvature, or both at an interval from the bottom hole with a length 200 approximately equal to that of the sonde.

#### 201 4. Testing stand and sonde prototype design

- 202 *4.1. Testing stand*
- 203 4.1.1. General testing stand design

The testing stand consists of a sledge, mast, top wheel, winch, and sonde prototype (Fig. 7). All stand parts are mounted on a sledge, which has a modular construction comprising a pair of skis and two welded frames bolted together. A 2-m-high mast is mounted in the middle of the sledge. A small winch is mounted near the mast on a sledge. A block is installed at the top of the mast. The testing stand parameters are listed in Table 5.

209 Table 5

### 210 Testing stand parameters

Mast height	2 m		
Max length of the testing sonde	~2 m		
Weight of the testing sonde	nom. 100 kg or less; max. 200 kg.		
Max. volume of the winch drum	10 m length of 5 mm Kevlar cable		
Min. ROP	0.1 m/h ( <i>ROP values refresh rate no more than once per ~6.5 sec</i> ) 0.72 m/h ( <i>ROP values 1 sec. refresh rate</i> )		
Max. possible tripping speed	9.3 m/min (weight of the testing sonde no more than 57 kg)		
Max. tripping speed	5 m/min (for 100 kg testing sonde) 2.3 m/min (for 200 kg testing sonde)		



213

Fig. 7. Schematic of the testing stand

# 214 *4.1.2. Top block and as-low-as-practicable ROP*

The top block consists of two wheels – a measuring wheel and a guiding wheel, an encoder, and two load-cell sensors. The measuring wheel is designed for a rope with a 5 mm diameter so that the cable length passing through the wheel per revolution equals  $1 \text{ m} \pm 1 \text{ mm}$ . This design simplifies the calculation and further adjustment of the measuring equipment. The guiding wheel is used to guide the rope from the winch to the measuring wheel.

To register the weight on bit (WOB), two load-cell sensors are installed underneath the measuring wheel assembly. Each sensor can withstand a force of up to 100 daN. To measure the ROP, an encoder with a resolution of 5000 measurements per revolution (MPR) is installed on the measuring wheel shaft. As the ROP is expected to be relatively low, the angular rotation speed of

- the measuring wheel is correspondingly small. Therefore, the higher the encoder resolution, the
- 225 more frequently it can capture instantaneous low-ROP values.

As-low-as-practicable ROP [m/h] can be estimated as follows:

$$\upsilon_{\min} = 3600 \cdot \pi \left( D + d \right) n_{\min} \tag{8}$$

227 where *D* is the wheel diameter (D = 0.3135 m); *d* is the rope diameter, and  $n_{\min}$  is the minimum angular

228 velocity in revolutions per second (RPS)

229 Minimal angular velocity  $n_{\min}$  is equal to:

$$n_{\min} = \frac{1}{tm} \tag{9}$$

230 where m is the encoder resolution in MPR and t is the time after which the data must be updated

231 (in this study, 
$$t = 1$$
 s).

The diameter of the rope axis on the wheel is:

$$D+d = \frac{l}{\pi} \tag{10}$$

where l is the rope length passing through the wheel per revolution.

After all rearrangements, the as-low-as-practicable ROP is:

$$\nu_{\min} = 3600 \cdot \frac{l}{tm} \tag{11}$$

Therefore, the as-low-as-practicable ROP, which can be measured with a 5000-MPR encoder,
1-m wheel circumference, and a measurement frequency of once per second, was 0.72 m/h.

237 *4.1.3. Winch* 

The winch is based on an RV50 series worm gearbox. For precise winch control, a 200-W power servo drive was chosen in this study. To compensate for the low servo power, a small PX60 series planetary gearbox with a gear ratio of 1:6 was installed between the worm gearbox and servo. The small drum was customised to hold one layer of 5-mm-diameter Kevlar rope with a length of 10 m. To simplify the winch construction, the drum was mounted directly on the output shaft of the worm reducer. Further details regarding the winch construction design are presented in Sup. 1.

#### 244 4.1.4. Control system

245 The control system consists of a box containing various data acquisition modules (Fig. 8). Data acquisition modules ADAM 4017+ and ADAM 4018+ were used to collect data from the 246 load-cell sensors and thermocouples, respectively. Counter-tachometer module CTA4001A was 247 248 used to receive and convert signals from the encoder on the measuring wheel. Two MIK-1100 249 modules were connected to voltage and current sensors. Temperature module DT320 was used to 250 monitor the sonde prototype thermal head temperature. All modules, along with the voltage and current sensors, were mounted in a BDH20016 black case. The wiring schematics for all 251 components are shown in Fig. S5.1 (Sup. 5). The sensor parameters are listed in Table 6. 252



#### 253 254

#### 255 Table 6

#### 256 Parameters of the sensors

Sensor type	Range	Accuracy	Mounting location	Meas. values	
Encoder BC58S10	up to 6000 RPM	5000 MPR	Top block	ROP, Depth	
Load-cell YZC-320C (2 pcs)	up to 100 kg	≤±0.02%	Top block	WOB	
Voltage sensor MIK-DJV-500V	up to 500V	0.2%	Control system box	Voltage	
Current sensor MIK-DJI-400A	up to 400A	1%	Control system box	Current	
T type thermocouples	from -270°C up to 370°C	±0.75%	Ice block, air, thermal head and control system box	Temperature	

### 257 **4.1.5. Software**

258 The control system box, servo control unit, and thermal head control unit were connected to a computer via RS-485. The MODBUS RTU communication protocol was used for data 259 transmission. The software registers the following parameters from the sensors connected to the 260 control system box: ROP (m/h) and Depth (m) (Section 2 in Fig. 9); WOB (daN) (Section 3); 261 Current (A), Voltage (V), Power (W), and three temperatures (°C) (Section 5). Through the 262 thermal head control unit, the software allows monitoring of the sonde prototype inclination and 263 heater status and allows selection between manual and automatic modes (Section 4). A control 264 panel for the winch is located at the bottom of the screen (Section 6). 265



**SECTION 1** 

- 266 267
- 268

269

4.2. Testing sonde prototype

270 *4.2.1. General structure of the sonde prototype* 

The sonde prototype consisted of a thermal head borrowed from the RECAS-200 prototype and a control unit assembled inside the housing (Fig. 10). The total sonde prototype length was

- approximately 1.1 m, and its weight was approximately 35 kg. The sonde prototype was suspended
- using a Kevlar rope tied to a hook. Electric lines for the power supply and communication were
- inserted through isolated connectors in the top cover.



276 277

Fig. 10. Schematic of the self-steering sonde prototype

279 *4.2.2. Thermal head* 

The thermal head diameter was 160 mm. It had 16 heat cartridges with a total power of approximately 7.6 kW (Li et al., 2020; Talalay et al., 2019). The heaters connections in the thermal head were redesigned (Fig. 11). Fuses were installed on each heater, and a distribution board was designed to distribute the load and connect it to the power connector (Fig. 12). To allow each heater to be individually connected to a power source, the power connector was also changed from two four-pin connectors to one 21-pin connector.





Fig. 12. Electrical schematic of thermal head

The thermal head uses eight long heaters (200 mm in length) and eight short heaters (150 mm in length) that were arranged in an alternating pattern. The cartridges were controlled in pairs, with each long heater paired with an adjacent short one (Fig. 12). Therefore, the number of required PWM signals (PWM channels) was reduced to eight. 

#### 295 *4.2.3. Control unit*

To control the heaters in the sonde prototype, a control unit was designed as a pressure chamber housing the following components (Fig. 13).







Fig. 13. Control unit: (a) schematic and (b) photo

Dual axis inclinometer. An inclinometer with two axes was chosen instead of a three-axis one 300 301 because the control unit was rigidly fixed together with the thermal head inside the sonde prototype, 302 eliminating the need to track the relative rotation along the vertical axis. Also, sonde rotation along 303 the axis will not change the steering direction, because the system constantly calculates new PWM coefficient. The task of tracking the borehole azimuth is planned for future RECAS prototype testing. 304 305 PWM generator for 20 channels (Sup. 2). Although only eight PWM channels were required 306 in this study, the PWM generator was designed with 20 channels to enable the control of individual heaters in the bottom thermal head in the RECAS prototype in the future. 307

308 *Two power modules, 4 channels each* (Sup. 3). The heater pairs were not connected directly 309 to the PWM generator but through power modules that amplify the corresponding PWM signals 310 from the generator.

311 *Two 15-volt DC-DC converters* (Sup. 4). Two identical DC-DC converters were used to 312 isolate the power supply of the inclinometer and the PWM generator from the power supply of the 313 low-voltage part of the power modules.

The wiring schematics for all components are shown in Fig. S5.2 (Sup. 5).

- All control unit modules, except for the inclinometer, were customised for this study. The
- 316 primary characteristics of the control units are listed in Table 7.

#### 317 Table 7

318 Control unit main parameters

Parameter	Value
Power supply	100-500 V DC
Limit values for angle	$X  ext{ axis } \pm 90$
measurement	Y axis ±45°
Angle measuring accuracy	0.2°
Number of PWM channels	8 (upgradeable to 20 Ch.)
Communication with PC	RS-485 MODBUS RTU

319 Further details regarding each individually designed module can be found in the

320 corresponding supplements.

### **5.** Laboratory testing of self-steering sonde prototype

- 322 5.1. Factors and main parameters of experiments
- 323 To study the sonde prototype inclination in a drilled borehole, a series of tests were
- 324 conducted in the Polar Research Center laboratory.
- 325 The main factors affecting the sonde inclination and drilling performance are:
- 326 1. Ice temperature (kept constant at  $-16^{\circ}$ C);
- 327 2. Environmental temperature (varied slightly between  $+7^{\circ}C$  and  $+12^{\circ}C$ );
- 328 3. ROP, which was controlled by the winch and limited by the power supplied to the heaters329 inside the thermal head;

4. WOB, which changed with the ROP and was limited by the sonde prototype weight.

The main parameter to be recorded was the sonde inclination. The sonde inclination was affected by the controlled directional heat distribution on the thermal head surface, which was controlled by limiting the heater power. The control algorithm, with two variable coefficients T and  $y_{off}$  is described in Eq. 3. Therefore, the main purpose of the experiments was to determine the dependence of the sonde inclination on these coefficients. For clarity and visual control, blocks

of transparent ice with dimensions of  $50 \times 50 \times 100$  cm were used in the experiments (Fig. 14).



- Fig. 14. Testing of the sonde prototype with RECAS-200 controllable thermal head
- 340 5.2. Preliminary experiment

344 345

A preliminary experiment was conducted to determine the potential ROP and WOB ranges and the possibility of controlling the sonde prototype inclination in the borehole by regulating the heater power in the thermal head. The experiment recording was divided into four sections (Fig. 15).



Fig. 15. Recording of the preliminary experiment

*Section 1 (0-6 min).* The first 200 mm of drilling were strictly vertical, with an ROP of 0.6 m/h and all heaters running at 50% power. Subsequently, the power was increased to the maximum, and half of the heaters on one side of the thermal head were switched off. Then, the ROP was increased to 1 m/h.

*Section 2 (6-11 min).* No sonde prototype inclination was detected, and the ROP was set to
2 m/h to increase the WOB.

*Section 3 (12-16 min).* The inclination began to increase rapidly when the WOB value reached approximately 25 daN. An attempt was made to stabilise the WOB at this value. The WOB stabilised at approximately 25 daN with an ROP of approximately 1.5 m/h.

Section 4 (16-30 min). When the sonde prototype inclination angle reached approximately 4°, the powered heaters configuration was changed. Four previously powered heater pairs were switched off, and four heater pairs on the opposite side were switched on. The inclination angle decreased to nearly zero and then gradually increased in opposite direction.

Based on the preliminary experimental results, the following conclusions can be drawn. To achieve the desired sonde prototype inclination, the WOB should be approximately 20 daN or higher. However, the test sonde weight was only 35 kg, which significantly reduced the range of the acceptable WOB values. To avoid a situation in which the entire sonde prototype weight would be at the bottom of the borehole, the WOB range was limited to 22-28 daN.

When a 50% power limit was applied, the WOB stabilised at an ROP of approximately 364 365 1.5 m/h. Although WOB is not directly controlled, it depends on the ROP. However, constant WOB 366 adjustments via ROP changes using a proportional-integral-derivative (PID) controller were not very effective because the transients significantly influenced the measured parameters. Therefore, in 367 368 subsequent experiments, we decided to maintain a constant ROP despite potential WOB fluctuations. Because the coefficient y<sub>off</sub> limits the maximum average PWM coefficient values, in practice, 369 370 it limits the power consumption of the thermal head, which, in turn, affects the maximum ROP. 371 Preliminary experimental results showed that testing was meaningful only at WOB values close to the maximum. This means that at a certain  $y_{off}$  value, it is not possible to change the ROP over a significant range. By analysing the "behaviour" of Eq. 3 we can conclude that at  $y_{off} = 0.5$ , the heat distribution is the most intense, and the assumed rate of change in borehole trajectory is also at its maximum. The limitation of the maximum borehole depth that can be obtained from the available ice blocks underscores its importance.

Based on the above, we conducted a series of experiments with four different correction coefficient *T* values. The ROP was kept constant at 1.5 m/h. The WOB stabilised between 22 and 28 daN. The power consumption was limited to 50% by setting  $y_{off} = 0.5$ .

According to the test plan, in the first approximately 300 mm of each experiment, the sonde prototype should drill with half of the heaters on one side turned off until the sonde inclination angle reaches approximately 4° (Ye et al., 2024). Subsequently, the automatic alignment mode will be enabled. The algorithm will recalculate the PWM coefficients of the heaters at 1-s intervals. It is worth noting that decreasing the PWM coefficient recalculation frequency (i.e., slowing the response to inclination angle changes) can influence the borehole deviation intensity. A decrease in the recalculation frequency is likely to result in a decrease in borehole deviation intensity.

### 387 5.3. Experimental results and discution

A total of four experiments were performed. The experimental recordings are shown in Fig. 16. As the values were recorded from the sensors at a frequency of once per second, the graphs are depicted with a 15-value moving-average filter. The graphs show trend lines for each experiment. For each trend line, the bold line indicates the accuracy limits according to the inclinometer specifications ( $\pm 0.2^{\circ}$ ).





Fig. 17. Approximations of four experiments

403 To understand how crucial borehole deviations are for the passability of the sonde prototype,

404 it is necessary to analyse the results for the allowable deviation intensity according to the method

described in Section 3. In the experiments, the automatic alignment length was approximately 0.5 m. For clarity, it was decided to divide this 0.5-m section of each borehole into several sections, for each of which the radius of curvature was determined. The option of partitioning the path not according to depth but rather according to inclination angle proved to be the most illustrative. Four sections were selected with the following inclination angle ranges:  $[4^{\circ}-2.5^{\circ}], [2.5^{\circ}-1.5^{\circ}], [1.5^{\circ}-1^{\circ}], and [1^{\circ}-0.5^{\circ}].$ 

The radius of curvature was determined for all four boreholes in each of the selected sections. To determine the radius, an additional angle (approximately in the middle of the range) was selected. The following additional angle values were selected for further calculations:  $3.25^{\circ}$  for the range [4°-2.5°], 2° for [2.5°-1.5°], 1.25° for [1.5°-1°], and 0.75° for [1°-0.5°]. Fig. 18 shows the sonde trajectory for the [4°-2.5°] range in borehole 2 (*T* = 1).



### 416 417

**Fig. 18.** Borehole 2 (T = 1) trajectory within inclination angle range of [4°-2.5°]

418 The sectional radius of curvature was calculated as follows:

$$R = \frac{abc}{4S} \tag{12}$$

419 where *a*, *b* and *c* are the side lengths of triangle *ABC* and *S* is the area of triangle *ABC*.

The area of the triangle can be determined using Heron's equation:

$$S = \sqrt{p(p-a)(p-b)(p-c)} \tag{13}$$

421 where *p* is the semi-perimeter of a triangle.

422 The side lengths of triangle *ABC* can be determined using the Pythagorean theorem given 423 the coordinates of the points  $A(x_1, z_1)$ ,  $B(x_2, z_2)$  and  $C(x_3, z_3)$  are known:

$$a = \sqrt{\left(x_2 - x_1\right)^2 + \left(z_2 - z_1\right)^2} \tag{14}$$

$$b = \sqrt{\left(x_3 - x_2\right)^2 + \left(z_3 - z_2\right)^2}$$
(15)

$$c = \sqrt{\left(x_3 - x_1\right)^2 + \left(z_3 - z_1\right)^2} \tag{16}$$

Knowing the coordinates of points *A*, *B*, and *C* for all sections using Eqs. 12-16, the radius of curvature of each section can be determined. Substituting the radius of curvature into Eq. 6, the maximum allowable sonde length satisfying the passability criteria for each segment can be determined. The resulting radius of curvature and maximum allowable sonde length values are presented in Table 8 and Fig. 19.

429 Table 8

420

#### 430 Radius of curvature and maximum allowable sonde length

Range	Borehole 1, $T = 0.5$		Borehole 2, $T = 1$		Borehole 3, $T = 1.5$		Borehole 4, $T = 2$	
	Radius of	Length of	Radius of	Length of	Radius of	Length of	Radius of	Length of
	curvature	sonde	curvature	sonde	curvature	sonde	curvature	sonde
	(m)	(m)	(m)	(m)	(m)	(m)	(m)	(m)
4°-2.5°	4.701	0.866	3.761	0.775	3.270	0.722	2.594	0.643
2.5°-1.5°	7.64	1.105	6.112	0.988	5.315	0.921	4.215	0.82
1.5°-1°	12.115	1.392	9.692	1.245	8.428	1.161	6.684	1.033
1°-0.5°	20.7	1.819	16.56	1.627	14.4	1.517	11.421	1.351



**432** Fig. 19. Dependence of the maximum allowable sonde length at different borehole sections with the same inclination change

It is worth noting that the difference between the initial and final angles has a significant impact on the radius of curvature. If the analysed section tends to zero, the radius of curvature of such a section tends to infinity, and vice versa. As a compromise, the ranges were selected to minimise the difference between the size of the deviation angles ranges and the corresponding borehole section lengths.

438 **6.** Conclusions

- Based on the experimental results for the sonde prototype, the main conclusions of this studycan be summarised as follows:
- 1. The sonde prototype demonstrates a promising potential in controlling the borehole direction
  and using the RECAS, it should be possible to control the borehole direction to a certain extent
  using the proposed method.
- 2. The borehole deviation intensity during drilling can be corrected by controlling the correctioncoefficient *T*.
- 3. It is worth noting that the radius of curvature of a real RECAS would be higher than that
  obtained experimentally. Further research is required to obtain the RECAS parameters.
  However, to prevent the RECAS from becoming stuck in its own borehole at the chosen
  experimental drilling parameters, the borehole deviation radius intensity must be reduced.
- 450 4. At the maximum borehole diameter value obtained in the field for the RECAS-200 prototype,

451 the maximum theoretical borehole deviation value cannot exceed  $0.67^{\circ}$  at a sonde length of

452 approximately 7 m. However, this calculation did not consider the fact that a 7-m-long sonde

- 453 may exhibit some deformability (especially at the joints), despite its housing being made of
- 454 stainless steel. At this length, even a small deformation of a few millimetres could positively
- 455 affect the passability of the sonde in the borehole.

456 5. Sonde passability at large borehole deviation intensity values can be improved if the housing457 is structurally divided into several parts capable of deviating from each other (hinged joints).

- 458 Allowing just a half-degree deviation of one part of the sonde from the other could increase its passability. 459
- In future work, we plan to conduct experiments on a larger scale (e.g., with a borehole depth 460 461 of approximately 10 m) to refine the results in a deviation intensity range closer to that obtained with a real RECAS. 462

#### Data availability 463

464

All raw data can be provided by the corresponding authors upon request.

#### **Author contributions** 465

Conceptualization: TPG, SMA, FX; hardware and equipment design: SMA, GD; software 466 467 development SMA; resources and supplies: FX, GD, DZ, ZN; planning the experiment: SMA, TPG; 468 assistance in preparing for experiments: FX, ZN, GD, DZ; conducting experiments and performed 469 the measurements: SMA, FX; analysing the data: SMA, TPG; project administration: FX, ZN; financial management: YY, WT; supervision TPG; writing the manuscript draft: SMA, TPG; 470 471 reviewing and editing the manuscript TPG; reviewing the manuscript: FX, GD.

#### **Competing interests** 472

The authors declare that they have no conflict of interest. 473

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